## A SYMPLECTIC INTEGRATION SCHEME THAT ALLOWS CLOSE ENCOUNTERS BETWEEN MASSIVE BODIES

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Mixed-variable symplectic integrators provide a fast, moderately accurate way to study the long-term evolution of a wide variety of N-body systems (Wisdom & Holman 1991). They are especially suited to planetary and satellite systems, in which a central body contains most of the mass. However, in their original form, they become inaccurate whenever two bodies approach one another closely. Here, I will show how to overcome this difficulty using a hybrid integrator that combines symplectic and conventional algorithms.

A symplectic integrator works by splitting the Hamiltonian, H, for an N-body system, into two or more parts  $H = H_0 + H_1 + \cdots$ , where  $\epsilon_i = H_i/H_0 \ll 1$  for  $i = 1, 2 \ldots$ . An integration step consists of several substeps, each of which advances the system due to the effect of one part of the Hamiltonian only. The error incurred over the whole step is  $\sim \epsilon \tau^{n+1}$ , where  $\tau$  is the timestep, n is the order of the integrator, and  $\epsilon$  is the largest of  $\epsilon_i$ .

A symplectic algorithm is efficient provided that the  $\epsilon$  factors are small. In the planetary system, this is usually achieved by making  $H_0$  the unperturbed Keplerian motion of the planets about the Sun, and  $H_1$ ,  $H_2$  etc. the perturbations between planets. For example, using mixed coordinates (heliocentric positions and barycentric velocities), H is split up as

$$H_{0} = \sum_{i=1}^{N} \left( \frac{p_{i}^{2}}{2m_{i}} - \frac{Gm_{\odot}m_{i}}{r_{i\odot}} \right)$$

$$H_{1} = -G \sum_{i=1}^{N} \sum_{j=i+1}^{N} \frac{m_{i}m_{j}}{r_{ij}}$$

$$H_{2} = \frac{1}{2m_{\odot}} \left( \sum_{i=1}^{N} \mathbf{p}_{i} \right)^{2}$$
(1)

where N is the number of planets, m denotes mass, and r, p are position and momentum respectively. Note that each part of the Hamiltonian can be integrated analytically in the absence of the others, and  $H_0$  contains all of the large terms, provided that the planets are widely separated.

Now consider a close encounter between bodies a and b. During the encounter, the distance  $r_{ab}$  is small, and the corresponding term in  $H_1$  is large. This means that  $\epsilon_1$  is no longer small and the integrator becomes inaccurate. An approximate solution to this difficulty is to transfer the offending term from  $H_1$  to  $H_0$  for the duration of the encounter. This ensures that  $\epsilon_1$  is always small. However, each time

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 $H_0$  and  $H_1$  are changed in this way, the system undergoes a shift in energy, and the integrator's symplectic property is lost.

A better solution is to split each of the interaction terms between  $H_0$  and  $H_1$  as follows

$$H_{0} = \sum_{i=1}^{N} \left( \frac{p_{i}^{2}}{2m_{i}} - \frac{Gm_{\odot}m_{i}}{r_{i\odot}} \right) - G \sum_{i=1}^{N} \sum_{j=i+1}^{N} \frac{m_{i}m_{j}}{r_{ij}} [1 - K(r_{ij})]$$

$$H_{1} = -G \sum_{i=1}^{N} \sum_{j=i+1}^{N} \frac{m_{i}m_{j}}{r_{ij}} K(r_{ij})$$
(2)

where the function K is chosen so that  $K \to 0$  when  $r_{ij}$  is small, and  $K \to 1$  when  $r_{ij}$  is large. This ensures that  $\epsilon_1$  is always small, without requiring that terms move from one part of the Hamiltonian to another.

When all of the separations  $r_{ij}$  are large,  $H_0$  can be advanced analytically as before (since 1 - K = 0). If two bodies undergo a close encounter, the terms in  $H_0$  due to these objects must be integrated numerically, but all the remaining terms can still be advanced analytically. By trial and error, I find that a good expression for K is

$$K = \begin{cases} 0 & \text{for } y < 0\\ y^2/(2y^2 - 2y + 1) & \text{for } 0 < y < 1\\ 1 & \text{for } y > 1 \end{cases}$$

where

$$y = \left(\frac{r_{ij} - 0.1 \, r_{crit}}{0.9 \, r_{crit}}\right)$$

and  $r_{crit}$  is the larger of 3 Hill radii and  $0.5\tau v_{max}$ , where  $v_{max}$  is the maximum likely orbital velocity of any of the objects.

## References

Wisdom J., Holman M.: 1991, Astron.J., 102, 1528.