

Study of centering CCD image of faint satellites near a bright primary object†

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Abstract. The polynomial-fit method is applied to remove the uneven background of a satellite when it is near a bright primary object. Detailed analysis of this method is given. Some useful conclusions are drawn from the results of simulated data.

Keywords. astrometry, solar system: general, techniques: image processing, planets and satellites: general

1. Introduction

When observing a faint satellite close to a bright primary object, the satellite is embedded in the primary's halo light, which has a gradient across the satellite's image. Usual methods, such as 2-dimension Gaussian with a constant background fitting, cannot provide a correct center of the satellite.

Many authors presented various methods to remove the systematic effects on centering the position of a satellite's image. But each method is only suitable for a special case. Here we attempt to find a general approach to tackle this problem. The "polynomial-fit" method has been mentioned and employed previously for removing uneven background without any detailed discussions or argued conclusions given. The goal of this work is to analyse which order of the polynomial is suitable for most cases.

2. Results from simulated data

First, the simulated images of a satellite near a bright primary were created under various assumptions, such as different distances between the satellite and the primary, different intensity ratios between the satellite and the primary, and so on. Second, different orders of the polynomial-fit method are used to fit the background when calculating the center of a satellite. Third, comparisons between the calculated and simulated centers are made.

Normally the images of both the satellite and the primary object can be represented by a two-dimension Gaussian model, with respectively different parameters of peak intensities I_s and I_p , center positions (x_s, y_s) and (x_p, y_p) , and Gaussian radius parameters R_s and R_p . The combined intensity of a satellite and a primary can be simulated as:

† Supported by the National Natural Science Foundation of China(Grant Nos. 10673026, 10333050, and 10573018) and Science & Technology Commission of Shanghai Municipality(06DZ22101).

$$I(x, y) = I_p \text{EXP}\left(-\frac{(x - x_p)^2 + (y - y_p)^2}{2R_p^2}\right) + I_s \text{EXP}\left(-\frac{(x - x_s)^2 + (y - y_s)^2}{2R_s^2}\right) + N \quad (2.1)$$

where N is the random error. To calculate the center of the satellite from the simulated data, a square area around the center of the faint satellite image is selected first. Then the following model is used to fit the center of the satellite.

$$I(x, y) = I_0 \text{EXP}\left(-\frac{(x - x_0)^2 + (y - y_0)^2}{2R_0^2}\right) + \sum_{i=0}^n \sum_{j=0}^n a_{ij} (x - x_0)^i (y - y_0)^j \quad (2.2)$$

where I_0 is the fitted intensity of the image of the satellite, (x_0, y_0) is fitted center of the satellite, and n is the order of the polynomial ($n=1, \dots, 5$). Here $I_0, R_0, (x_0, y_0)$ and a_{ij} are fitted by least squares.

Since the correct positions of the satellite are known when simulated under different orders of polynomial $n(n = 1, 2, 3, 4, 5)$, the mean difference between the simulated and calculated center positions (under different random errors) can be used to judge the validity of different orders of the polynomial. Table 1, 2 and 3 respectively show the simulation results for various values of $I_p/I_s, r_{ps}$ and R_p .

Table 1. Results of five models with different (I_p/I_s) under different r_{ps}/R_p when $R_p=32$ pixel. Note: "×" means the result is not convergent with this model, "Δ" means the residual is very large with this model, and "○" means good results for this model.

model	I_p/I_s ($r_{ps}/R_p=2$)			I_p/I_s ($r_{ps}/R_p=3$)			I_p/I_s ($r_{ps}/R_p=4$)
	(12.5~25)	(25~37.5)	(37.5~250)	(12.5~25)	(25~100)	(100~250)	(1.0~250)
1-deg	×	×	×	Δ	Δ	×	○
2-deg	Δ	Δ	Δ	○	○	×	○
3-deg	○	Δ	Δ	○	Δ	Δ	○
4-deg	○	○	Δ	○	○	○	○
5-deg	○	○	○	○	○	○	○

Table 2. Results of five models with different r_{ps}/R_p under different R_p (unit: pixel).

model	r_{ps}/R_p ($R_p=8$)			r_{ps}/R_p ($R_p=16$)			r_{ps}/R_p ($R_p=32$)		
	(1~4)	(4~5)	(>5)	(1~3)	(3~4)	(>4)	(1~2)	(2~3)	(>3)
1-deg	×	×	○	×	Δ	○	×	×	○
2-deg	×	×	○	×	Δ	○	Δ	○	○
3-deg	Δ	Δ	○	○	○	○	○	○	○
4-deg	Δ	○	○	○	○	○	○	○	○
5-deg	Δ	○	○	○	○	○	○	○	○

3. Discussion

Synthetically produced star images have been used to investigate the properties of the polynomial-fit method. Five polynomial models have been tested under a wide range of

Table 3. Results of five models with different R_p (unit:pixel) under different r_{ps} (unit: pixel). Here $I_p/I_s = 25/8$

model	R_p (when $r_{ps}=30$)			R_p (when $r_{ps}=50$)			R_p (when $r_{ps}=80$) (10~34)
	(10~12)	(12~16)	(16~34)	(10>12)	(12~14)	(14~34)	
1-deg	×	×	×	○	○	×	○
2-deg	×	×	×	○	△	△	○
3-deg	○	×	×	○	○	○	○
4-deg	△	△	○	○	○	○	○
5-deg	○	○	○	○	○	○	○

observational conditions. The numerical calculations show that the polynomial-fit method is useful for correcting the effects of the strong halo light while centering the image of a faint satellite. In addition, the applicable range of the 1st-order polynomial model is limited, while the 5th-order polynomial model produces the highest precision. Usually, the 2nd, 3rd and 4th-degree polynomials can also satisfy the requirements of accuracy level for actual observations. After obtaining the approximate values of I_p/I_s , r_{ps}/R_p , and R_p , we can determine which model is suitable with the help of Tables 1-3.

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